

Math 356CL

Homework #2

1) W+H 3-22

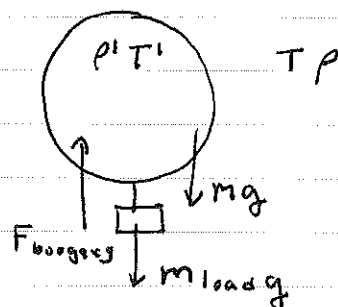
$$V_{\text{balloon}} = 3000 \text{ m}^3$$

$$M_{\text{load}} = 600 \text{ kg}$$

$$T_{\text{ground}} = 20^\circ\text{C}$$

$$\Gamma = 0$$

$$p = 900 \text{ hPa}$$



If $\Gamma = 0$ $\frac{dT}{dz} = 0 \rightarrow$ Temperature does not change with height. Then $T = 20^\circ\text{C}$ at 900 hPa

$$F_{\text{buoyancy}} - m_{\text{load}}g - mg = 0$$

$$\rho'Vg - m_{\text{load}}g - \rho Vg = 0$$

$$p = R_d \rho T$$

$$p = R_d \rho' T' \quad \text{Since } p' = p$$

$$\rho T = \rho' T' \quad \rho' = \frac{\rho T}{T'}$$

$$\rho' V = \rho V - m_{\text{load}}$$

$$\rho \frac{T}{T'} V = \rho V - m_{\text{load}}$$

$$\frac{1}{T'} = \frac{1}{T} \left(\rho V - m_{\text{load}} \right)$$

$$\frac{1}{T'} = \frac{1}{T} \left(1 - \frac{m_{\text{load}}}{\rho V} \right) \quad T' = \frac{T}{\left[1 - \frac{m_{\text{load}}}{\rho V} \right]}$$

and pressure 500 hPa

To get the density of air at 20°C we must use the information at STP ($T^\circ\text{C}$, $p = 1000 \text{ hPa}$, $\rho = 1.275 \frac{\text{kg}}{\text{m}^3}$)

$$\frac{p_1 T_1}{\rho_1} = \frac{p_2 T_2}{\rho_2} \rightarrow \rho_2 = \rho_1 \left(\frac{T_1}{T_2} \right) \left(\frac{p_2}{p_1} \right)$$

$$\rho = \rho_2 = \left(1.275 \frac{\text{kg}}{\text{m}^3} \right) \left(\frac{273}{293} \right) \left(\frac{500}{1000} \right)$$

$$\rho = 1.069 \text{ kg/m}^3$$

• Now substitute back into the equation

$$T' = \frac{293 \text{ K}}{\left[1 - \frac{300}{(1.069)(3000)} \right]} = 360.4 \text{ K}$$

$$T' = 360.4 - 273 = 87.4^\circ\text{C}$$

2) W44 3.37

$$p = 200 \text{ hPa} \quad T = -60^\circ\text{C} = -60 + 273 = 213 \text{ K}$$

Adiabatic compression to 1000 hPa

$$pV^\gamma = \text{constant for adiabatic process} \quad \gamma = 1.4$$

$$\frac{pV}{T} = \text{constant for ideal gas}$$

$$V \propto \frac{T}{p}$$

$$p \left(\frac{T}{p} \right)^\gamma = \text{constant}$$

$$p^{1-\gamma} T^\gamma = \text{constant}$$

$$T_2 = \left(T_1^\gamma \frac{p_1^{1-\gamma}}{p_2^{1-\gamma}} \right)^{\frac{1}{\gamma}} = T_1 \left(\frac{p_1}{p_2} \right)^{\frac{1-\gamma}{\gamma}} = 213 \text{ K} \left(\frac{200}{1000} \right)^{\frac{(1-1.4)}{1.4}}$$

$$T_2 = 337.3 \text{ K}$$

$$T_2 = 64.4^\circ \text{C}$$

$$3) \quad \underline{w + H \quad 3.41}$$

$$q = 0.0196$$

$$T = 30^\circ \text{C} = 303 \text{ K}$$

$$p = 1013 \text{ hPa}$$

$$q = \frac{m_v}{m_v + m_d} = \frac{w}{1 + w}$$

$$w = \frac{m_v}{m_d} \quad \text{mixing ratio}$$

• Virtual temperature

$$T_v = T \frac{1}{\left(1 - \frac{e}{p}(1 - \epsilon)\right)}$$

We need to find the vapor pressure e

$$e = \frac{n_v}{n_v + n_d} p = \frac{\frac{m_v}{M_w}}{\frac{m_v}{M_w} + \frac{m_d}{M_d}} p =$$

$$e = \frac{1}{m_d M_w} \left(\frac{m_v}{m_d M_w} + \frac{1}{M_d} \right) p = \frac{w}{(w + \epsilon)} p \quad \epsilon = \frac{M_w}{M_d} = 0.622$$

→ We can solve for w from the specific humidity

$$q(1 + w) = w$$

$$q = w(1 - q)$$

$$w = \frac{q}{(1 - q)}$$

$$w = \frac{0.0196}{(1 - 0.0196)} = 0.02 \quad \leftarrow \text{Appears without units}$$

We must assume
the are the same in
numerator & denominator

$$w = 0.02 \frac{\text{kg}}{\text{kg}} = 20 \frac{\text{g}}{\text{kg}}$$

$$T_v = \frac{T}{\left(1 - \left(\frac{w}{w+e}\right)(1-\epsilon)\right)} = \frac{303 \text{ K}}{\left(1 - \left(\frac{0.02}{0.02+0.622}\right)(1-0.622)\right)}$$

$$T_v = 306.6 \text{ K} = 33.6^\circ\text{C}$$

• Density

$$p = R_d \rho_{\text{moist}} T_v$$

$$\rho_{\text{moist}} = \frac{p}{R_d T_v} = \frac{1013 \times 10^2 \frac{\text{N}}{\text{m}^2}}{\left(287 \frac{\text{J}}{\text{K}}\right)(306.6 \text{ K})} =$$

$$\rho_{\text{moist}} = 1.15 \frac{\text{kg}}{\text{m}^3}$$

4) W+H 3.42

$$p = 975 \text{ hPa}$$

$$T = 15^\circ\text{C} = 288 \text{ K}$$

$$w = 1.8 \frac{\text{g}}{\text{kg}} = 1.8 \times 10^{-3} \frac{\text{kg}}{\text{kg}} = 1.8 \times 10^{-3}$$

From the derivation in the last problem

• Vapor pressure

$$e = \frac{w}{w+e} p = \frac{(1.8 \times 10^{-3})}{(1.8 \times 10^{-3} + 0.622)} (975 \text{ hPa})$$

$$e = 2.81 \text{ hPa}$$

• Virtual Temperature

$$T_v = \frac{T}{\left(1 - \frac{e}{p}(1-\epsilon)\right)} = \frac{288}{\left(1 - \frac{2.81}{975}(1-0.622)\right)} = 288.3 \text{ K}$$

$$= 15.3^\circ\text{C}$$

5) W + H 3.5rIn 1st Carnot Cycle $T_1 = 100^\circ\text{C}$, $T_2 = 0^\circ\text{C}$ $Q_1 = 20 \text{ J}$ in one cycle

$$\frac{Q_1}{T_1} = \frac{Q_2}{T_2} \rightarrow Q_2 = Q_1 \left(\frac{T_2}{T_1} \right)$$

$$W = Q_1 - Q_2 = Q_1 \left[1 - \frac{T_2}{T_1} \right]$$

$$W = 20 \text{ J} \left[1 - \frac{273}{373} \right] = 5.36 \text{ J for one cycle}$$

$$W_{\text{total}} = 10 (5.36) = 53.6 \text{ J}$$

6) W + H 3.6f $P_{\text{top}} = 600 \text{ hPa}$ $P_{\text{bottom}} = 1013 \text{ hPa}$

$$\alpha_v = \frac{1}{\rho_v} = 1.66 \quad \frac{\text{m}^3}{\text{kg}} \quad \alpha_w = \frac{1}{\rho_w} = 1.00 \times 10^{-3} \frac{\text{m}^3}{\text{kg}}$$

• From eq 3.99

$$\frac{\Delta T_0}{\Delta P_{\text{atmos}}} = \frac{T_0 (\alpha_v - \alpha_w)}{L_v} \quad L_v = 2.25 \times 10^6 \frac{\text{J}}{\text{kg}}$$

$$T_0 = 100^\circ\text{C} = 373 \text{ K}$$

$$\Delta T_0 = \frac{T_0 (\alpha_v - \alpha_w) (P_{\text{top}} - P_{\text{bottom}})}{L_v}$$

$$\Delta T_0 = \frac{(373) (1.66 - 1.00 \times 10^{-3}) (600 - 1013) \times 10^2}{2.25 \times 10^6} \text{ K}$$

$$\Delta T_0 = -11.3 \text{ K} = -11.3^\circ\text{C}$$

$$T_0' = T_0 + \Delta T_0 = 100 - 11.3 = 88.7^\circ\text{C}$$

$$\Delta T = (T_2 - T_1) = (T_2 + 273 - (T_1 + 273))$$

(Notice that a difference in temperature is independent of scale)